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# Validation of a Digital Noise Power Integration Technique for Radiometric Clear Sky Attenuation Estimation at Q-Band

Alexios Costouri, James Nessel, George Goussetis

**Abstract**—This paper presents the validation of a digital signal processing technique that can be used to estimate radiometric sky noise, and hence, atmospheric absorption, within existing digital receivers at little/no additional cost. To demonstrate this, a receiver was constructed that simultaneously records the beacon signal power from the ALPHASAT Aldo Paraboni technology demonstration payload, as well as the integrated noise power in the adjacent band. Calibration from the digital radiometer is performed using tip-curve calibration procedures. Atmospheric fading is then obtained by observing the beacon as well as the radiometric signals. This enables the comparison of fading obtained by the two techniques and provides a means to calibrate the received beacon power level to obtain total atmospheric attenuation. It is shown that for low levels of fading, up to a few dB, the two techniques provide good agreement. This approach can therefore provide a low-cost option for geostationary mm-wave satellite channel measurements in the low fading regime, which can be useful in the design and operation of the feeder links in emerging satcom systems.

**Index Terms**—ALPHASAT, calibration, propagation, Q-band, radiometer, software-defined radio(SDR), tip-curve.

## I. INTRODUCTION

THE trends towards broadband services, high-throughput satellites and reduction of the per MBps cost drives the use of ever higher frequencies in satellite communications. While the use of Ka-band is now well established, there is an increasing interest in higher bands such as Q/V bands (40/50 GHz) [1] and W-band [2]. Broad bandwidths available at these frequency bands offer attractive opportunities for exploitation in the feeder links, since they can support high capacity with fewer gateways. Consequently, the cost of the ground segment is reduced while the entire Ka-band becomes available for revenue-raising user links [3]. Aforementioned developments motivate the characterisation of atmospheric propagation effects and their properties at mm-wave frequencies. An early campaign to experimentally characterise atmospheric propagation effects beyond the Ka-band was made with the ITALSAT F1 satellite which carried beacons covering Europe at three bands, Ka, Q and V band (20, 40 and 50 GHz respectively) [4],

[5]. Presently, the Aldo Paraboni payload hosted on-board the ALPHASAT satellite delivers beacons with similar coverage at Ka and Q band (19.701, 39.402 GHz) underpinning an ongoing Europe-wide test campaign [6], [7]. Anticipating future needs for characterisation at W-band, ESA is currently making preparations for a beacon at these frequencies in geostationary orbit [2].

While beacons on-board geostationary satellites provide a very accurate methodology to characterise the channel from this orbit, such missions are costly and typically require long developments prior to launching. To this end alternative methodologies are currently explored such as the characterization of W-band propagation using cubesats [8] or from purely ground-based observations [9]. In particular, passive cosmic background radiometry represents a popular methodology to characterise atmospheric fading [10]. This technique is based on measuring received noise from a portion of the sky without radio emitters. By virtue of calibrating the received noise, which is a combination of atmospheric emission and cosmic background radiation, atmospheric fading can be estimated without relying on a spaceborne beacon. This methodology delivers its maximum accuracy when the effects of larger atmospheric particles can be neglected, e.g. in the absence of precipitation.

The primary aim of this paper is to experimentally evaluate the potential of a software defined radio (SDR) receiver to deliver aforementioned passive radiometry measurements and characterize the geostationary channel at the satellite downlink without disrupting nominal operations. This is achieved as a result of measurements performed with a bespoke terminal designed to receive the Q-band signal from the beacon of the ALPHASAT Aldo Paraboni payload [11]. By virtue of the SDR setup, where a downconverted image of the received signal is sampled and digitally processed to obtain its spectrum, it is possible to simultaneously record the level of the intended signal (in this case, a CW beacon), as well as the noise in the nearby frequency band. The specific configuration enables an accurate comparison of the atmospheric attenuation derived from the radiometer, with that obtained from the beacon observation. Earlier literature such as [12] and [13], concurrently recorded the beacon satellite signal and noise power at Ka and Q band in which the calibration of the radiometer in post-processing required the satellite beacon measurements. Furthermore, in [14] and [15], the concurrent recordings helped to monitor events and obtain reference levels to estimate attenuation templates. The proposed configuration

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A. Costouri and G. Goussetis are with School of Engineering & Physical Sciences; Institute of Sensors, Signals and Systems, Heriot-Watt University, Edinburgh, EH14 4AS, Scotland, UK (e-mail: ac312@hw.ac.uk, g.goussetis@hw.ac.uk).

J. Nessel is with the NASA Glenn Research Center: Advanced High Frequency Branch, Cleveland, Ohio, USA (e-mail: james.a.nessel@nasa.gov).

provides a potential method for operational ground stations to monitor fading in real time with no additional hardware required.

The remainder of this paper is organised as follows. Section II outlines the theory underpinning the radiometric measurement and calibration. Section III presents some details of the system and the performed calibrations. Section IV presents the comparative experimental results between fading obtained from the beacon and the radiometric measurements. Finally, conclusions are presented in Section V.

## II. PRINCIPLE OF OPERATION

An electromagnetic signal that propagates through the atmosphere experiences attenuation due to absorption and scattering. Excluding free space losses (FSL), the signal attenuation along a path (also referred to as the extinction of the signal) can be obtained as [10]:

$$A_t = A_{ab} + A_{sc} \quad (dB) \quad (1)$$

where  $A_t$  is the total attenuation along the signal path,  $A_{ab}$  is the attenuation due to absorption and  $A_{sc}$  is the attenuation due to scattering.

The attenuation due to absorption is attributed to the interaction of electromagnetic energy with the gaseous molecules (primarily oxygen and water vapour in the 1-100 GHz range). At thermal equilibrium, energy transitions of these particles lead to emission and absorption of electromagnetic energy at equal rates. With this assumption, a signal propagating along a given path in the atmosphere will therefore experience atmospheric (or gaseous) absorption at the same rate [10]. Gaseous absorption dominates fading when, in the radiative transfer equation, the effects of scattering or other absorptive mechanisms of electromagnetic wave from atmospheric particles can be neglected [10]. This typically applies in e.g. the absence of precipitation. In this case, gaseous absorption can be characterised by measuring the received noise power considering the thermodynamics of the atmosphere as briefly outlined next.

Commencing from the radiative transfer equation and assuming that the atmosphere can be considered in thermal equilibrium, it can be shown that the atmospheric absorption can be approximated as [10]:

$$A_{ab} = 10 \log_{10} \frac{T_m - T_c}{T_m - T_B} \quad (dB) \quad (2)$$

In (2)  $T_B$ ,  $T_m$  and  $T_c$  correspond to the brightness, mean brightness and cosmic radiation temperatures respectively. The brightness temperature,  $T_B$ , represents the temperature of a black body in thermal equilibrium that would have the same brightness at the frequency of interest as the isotropic atmosphere [10]. The term  $T_m$ , stands for the mean radiating temperature of the atmosphere/absorbing medium [10]. Available guidelines (ITU-R P.618-13, [16]) provide empirical expressions to obtain  $T_m$  as a function of the surface temperature,  $T_s$ , for clear and cloudy weather, as in (3) below.

$$T_m = 37.34 + 0.81T_s \quad (K) \quad (3)$$

The cosmic radiation temperature,  $T_c$ , accounts for the contribution from the cosmic radiation associated with the remnants from the Big Bang at the mm-/micro-wave range and has a constant value of 2.7K [17].

The above suggests that the gaseous absorption of the atmosphere can be estimated once values of the atmosphere's brightness temperature,  $T_B$ , are available. The latter can be obtained exploiting radiometric noise measurements. In particular, a lossless antenna enclosed in a black body in thermal equilibrium (which with the above approximation can be assumed to be the atmosphere) will detect the following noise power [17]:

$$P_{N*} = kT_BB \quad (W) \quad (4)$$

where,  $P_{N*}$  is the noise power received by the antenna over a certain bandwidth  $B$ , with  $k$  being the Boltzmann's constant and  $T_B$  the brightness temperature in Kelvin. Consequently, the noise temperature of a clear sky is proportional to the noise power recorded in an ideal receiver comprising a lossless antenna and noise-free electronics.

For practical receivers additional considerations are required. The RF electronics generate noise that can be calculated from the noise figure of each component, which can then be converted to an equivalent noise temperature,  $T_r$  [18]. Furthermore, a practical antenna is not lossless but instead is characterized by a noise temperature,  $T_N$ . Moreover, in addition to the main lobe (which observes the desired brightness temperature,  $T_B$ ) the antenna also has side lobes that capture noise from unwanted sources and the contributions are represented by  $T_{sl}$ . The total gain,  $G$ , of the receiver system must also be taken into account since gain applies to both the noise and desired signal. With the above considerations, (4) for a lossy receiver system becomes [17]:

$$P_N = (T_B + T_{sl} + T_r + T_N)kBG \quad (W) \quad (5)$$

Rearranging the equation to brightness temperature:

$$T_B = \frac{P_N}{kBG} - (T_{sl} + T_r + T_N) \quad (K) \quad (6)$$

Recognising the dependence of the brightness temperature,  $T_B$ , on the measured noise power,  $P_N$ , (6) can be written in the format:

$$T_B = aP_N + b \quad (K) \quad (7)$$

where, the coefficients  $a$  and  $b$  stand for:

$$a = \frac{1}{kBG} \quad (KJ^{-1}Hz^{-1}) \quad (8)$$

$$b = -(T_{sl} + T_r + T_N) \quad (K) \quad (9)$$

While approximate estimations for the coefficients  $a$  and  $b$  can be obtained by (8) and (9), more accurate values can be obtained by performing experimental calibration to remove the system's gain, noise and antenna side lobe contributions from the recorded noise power. A tip calibration exploits the fact that directions closer to the horizon experience longer paths through the atmosphere and consequently are associated with higher attenuation (and thus atmospheric noise) levels [19]. On the assumption that the atmosphere is homogeneous

with constant density, the attenuation,  $A_{ab}$ , at a given elevation angle,  $\theta$ , is proportional to the optical path length at that angle normalised to that at zenith. This term is also known as air mass and for elevation angles over  $10^\circ$ , where the Earth's curvature can to a good approximation be neglected, is obtained from simple trigonometry [10], [20]:

$$\text{Air mass} = \frac{1}{\sin(\theta)} = \csc(\theta) \quad (10)$$

On that basis, the attenuation  $A_{ab}$  at angle  $\theta$  is linked with the attenuation  $A_{ab90}$  at zenith ( $\theta = 90^\circ$ ) according to:

$$A_{ab} = A_{ab90} \csc(\theta) = A_{ab90} * \text{Air mass} \quad (\text{dB}) \quad (11)$$

Eq. (11) reveals that the relationship between airmass and attenuation is linear and the associated curve (also referred to as tip curve) crosses the origin. It is noted that although an air mass lower than unity is practically impossible, theoretically this extrapolation is meaningful. In particular, air mass of zero corresponds to absence of the atmosphere and therefore no attenuation of a signal [17].

During tip calibration the antenna is pointed at different elevation angles, corresponding to different values for the air mass, and the noise power is recorded. An initial estimation of the zenith attenuation,  $A_{ab90}$  in dB, can be calculated by comparing two elevation angles [19].

$$A_{ab90} = \frac{4.343}{\csc(\theta) - 1} \ln \frac{T_{B90} - T_m}{T_B(\theta) - T_m} \quad (\text{dB}) \quad (12)$$

where,  $T_{B90}$  and  $T_B(\theta)$  are the brightness temperature at zenith and at an elevation angle  $\theta$  respectively. The temperatures  $T_{B90}$ ,  $T_B(\theta)$  and  $T_m$  in (12) are unknown. Given that, as per (7), the relationship between measured noise power and temperature is linear, (12) can be rewritten in terms of the associated integrated noise power levels.

$$A_{ab90} = \frac{4.343}{\csc(\theta) - 1} \ln \frac{P_N(T_{B90}) - P_N(T_m)}{P_N(T_B(\theta)) - P_N(T_m)} \quad (\text{dB}) \quad (13)$$

To a first approximation, the mean radiating temperature,  $T_m$ , can be replaced with that of the ambient air temperature  $T_a$ . In turn, the integrated noise power associated with the ambient temperature,  $P_N(T_a)$ , can be replaced with the observation of the brightness of a blackbody at ambient temperature. Therefore, (13) can be re-written as:

$$A_{ab90} = \frac{4.343}{\csc(\theta) - 1} \ln \frac{P_N(T_{B90}) - P_N(T_a)}{P_N(T_B(\theta)) - P_N(T_a)} \quad (\text{dB}) \quad (14)$$

and a first estimate of the calibration coefficients can be made as:

$$a_{est} = \frac{e^{A_{ab90}/4.343} (T_c - T_a)}{P_N(T_{B90}) - P_N(T_a)} \quad (\text{KJ}^{-1}\text{Hz}^{-1}) \quad (15)$$

$$b_{est} = T_a - a_{est} (P_N(T_a)) \quad (\text{K}) \quad (16)$$

Next we consider that the mean brightness temperature differs from the black body observed at ambient temperature

$$\Delta T = T_a - T_m \quad (\text{K}) \quad (17)$$

Consequently, the initial estimation of  $a_{est}$  can be used to find more accurate values of  $P_N(T_m)$  as:

$$P_N(T_m) = P_N(T_a) - \frac{\Delta T}{a_{est}} \quad (\text{W}) \quad (18)$$



Fig. 1. Photograph of the receiver.

The updated calibration coefficients,  $a_{est*}$  and  $b_{est*}$ , and updated atmospheric attenuation at zenith,  $A_{ab90*}$ , can then be obtained by introducing (18) into (14)-(16):

$$A_{ab90*} = \frac{4.343}{\csc(\theta) - 1} \ln \frac{P_N(T_{B90}) - P_N(T_m)}{P_N(T_B(\theta)) - P_N(T_m)} \quad (\text{dB}) \quad (19)$$

$$a_{est*} = \frac{e^{A_{ab90*}/4.343} (T_c - T_a - \Delta T)}{P_N(T_{B90}) - P_N(T_m)} \quad (\text{KJ}^{-1}\text{Hz}^{-1}) \quad (20)$$

$$b_{est*} = T_a - \Delta T - a_{est*} (P_N(T_a)) \quad (\text{K}) \quad (21)$$

In a tip calibration process, the calibration coefficients are estimated for the original approximation  $T_m = T_a$ . Based on this estimation, the tip curve (i.e. attenuation vs. airmass) is then plotted. Typically, the original tip curve does not cross the origin due to the non-zero value of  $\Delta T$  in (17). To this end, an initial assumption of  $\Delta T$  is made and the updated calibration coefficients,  $a_{est*}$  and  $b_{est*}$  are calculated. Rapid convergence of the calibration coefficients is achieved by iterating (18)-(21) with the assumed  $\Delta T$ . The latest calculated calibration coefficients are used in each iteration. The correct calibration coefficients are calculated by adjusting  $\Delta T$  until the tip-curve plot extrapolates to zero.

### III. SDR RECEIVER

The opportunities and limitations to characterise fading using the aforementioned radiometric approach at the Q-band satellite downlink frequency using a low-cost SDR receiver are next investigated by means of an experimental setup. The latter involves a Q-band receiver mounted on a pointing system shown in Fig. 1 that tracks the ALPHASAT satellite using an

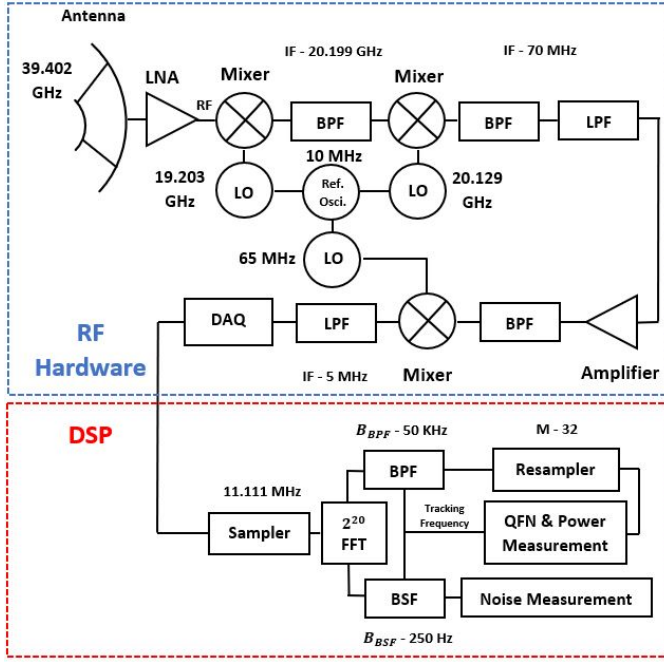


Fig. 2. Beacon receiver RF hardware (top) and DSP (bottom) block diagram.

open-loop system based on ephemeris data. The remainder of this section provides a description of the SDR receiver and its calibration.

#### A. System Description

The RF and digital signal processing (DSP) block diagram of the SDR Receiver is shown in Fig. 2. The receiver captures a linearly polarised signal at Q-band which is down-converted to an intermediate frequency (IF) of 5 MHz exploiting three-stage mixing. An ultra-stable 10 MHz reference oscillator is used across the 3 down conversion stages. The hardware is in a thermally insulated housing whose internal temperature is recorded. The signal is digitized at the 5 MHz IF frequency using a 12-bit National Instruments 5124 data acquisition (DAQ) card, which samples at a rate of 11.111 MHz.

The time-domain sampled signal is converted to the frequency domain exploiting a Fast Fourier Transform (FFT) on  $2^{20}$  time domain samples, leading to a resolution of 10.6 Hz. Then a digital 50 kHz, 10th order Type 2 Chebyshev bandpass filter (BPF) is applied to isolate the beacon signal and the resulting spectrum is digitally resampled by a factor of 32 to reduce processing time as shown in Fig. 2. A novel digital routine enables tracking any frequency drift of the beacon signal and together its peak power level [21]. Moreover, the Quinn-Fernandes Nessel (QFN) routine centres the filter accordingly. This routine is a variant of the Quinn-Fernandes frequency estimator [22] which uses *a priori* information window on the frequency that the beacon is expected to appear, resulting in faster detection. This allows to minimize scalloping losses as the QFN estimator interpolates the FFT and the total power of the beacon is recorded.

A typical image of the decimated spectrum detected by the SDR receiver is shown in Fig. 3. The received beacon

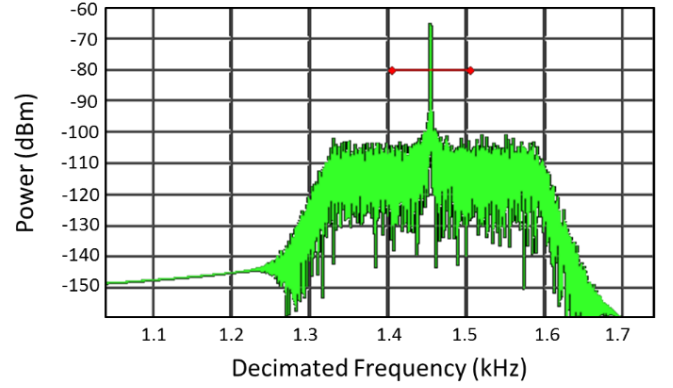


Fig. 3. Decimated IF spectrum with beacon signal present.

power is recorded at 10 Hz and 1 Hz. These are then used in conjunction with atmospheric data provided by European Centre for Medium-Range Weather Forecasts (ECMWF) vertical profiles and weather station data to produce a reference level such that the gaseous absorption and excess attenuation can be estimated [13]. The ECMWF data are available for every 6 hours on a regular latitude/longitude grid with  $0.125^\circ \times 0.125^\circ$  spatial resolution. The ECMWF vertical profile data (temperature, humidity, pressure) are processed with the well-known mass absorption models by Liebe et al. [23] and Rosenkranz [24] and the total path attenuation composed of oxygen, water vapour and cloud path attenuation are thereby obtained for the site. Identification of non-rainy periods is made from the weather station data, and in conjunction with the ECMWF data, a daily averaged gaseous attenuation level can be estimated. Once the reference attenuation level has been calculated for each day, it is subtracted from the measured power time series to produce the total atmospheric attenuation (including gaseous losses). Example of calibrated time series with the above approach can be seen in [11], [25] and [26]. A more detailed description of the receiver can be found in [13].

A digital 10th order Type 2 Chebyshev bandstop filter (BSF) with a rejection bandwidth,  $B_{BSF}$ , of 250 Hz, is concurrently applied on the spectrum resulting from the FFT as shown in Fig. 2. The centre of the digital bandstop filter is defined by the QFN frequency estimation of the beacon signal. This operation enables to virtually suppress the beacon signal from the received spectrum and hence allowing the integration of the noise power measurement over the full IF bandwidth of 1 MHz. A Type 2 Chebychev filter, unlike a Type 1, has a flat bandpass response. An instance of the bandstop filter response is plotted in Fig. 4. It is noted that the Aldo Paraboni beacon signal is a continuous wave signal (not modulated) with low phase noise (theoretically no bandwidth) hence the choice of the narrow bandwidth notch filter. The integrated noise power is logged with the same frequency as the beacon signal.

The capability of this receiver to perform as a low-cost radiometer is next demonstrated by means of a hot-cold target experiment. A Radar Absorbing Material (RAM) with sharp 5mm tall pyramidal and total tile dimensions of 100x100mm injection moulded conductive plastic, with reflection below



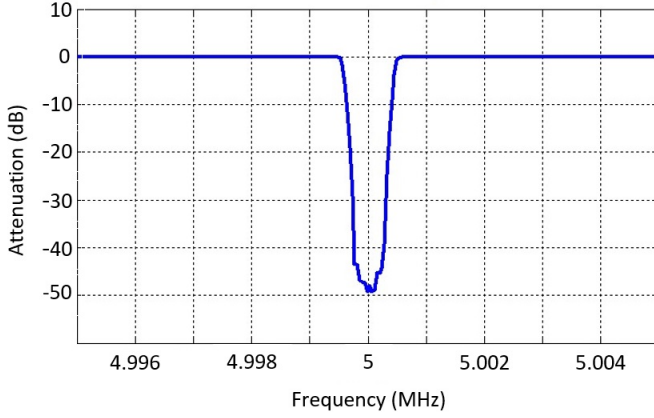


Fig. 4. Digital notch filter response for measuring the noise power of the beacon.

45dB at 40 GHz, was used as a black body target [27]. A cuboid box was built to host the RAM absorber using Rohacell (relative dielectric permittivity  $\epsilon_r = 1.05$ ) and hence largely RF transparent. The box was of sufficient dimensions to be accommodated on the dish while the RAM absorber fully covered the antenna feed. Throughout the experiment, the antenna was pointed to zenith for easy access as well as providing a level plane for the calibration box to rest upon, Fig. 5. During the experiment, the integrated noise power was measured for two different temperatures of the black body target, in particular the ambient air during the procedure (hot temperature measured at 280K) and liquid nitrogen (cold temperature approximately at 77K). The ambient air temperature during the calibration was recorded from a nearby weather station (building south of the receiver at a distance of approximately 70m) at 1-minute intervals.

In order for the absorber to first acquire the ambient (hot) temperature, it was left outdoors under shade for an hour. The absorber was then placed at the bottom of the box and the first measurement was recorded. In order to then obtain the cold source measurement, liquid nitrogen was introduced in the host Rohacell box. During the first attempt it was not possible for the absorber to acquire the cold temperature as it started floating in the liquid nitrogen. A second attempt was made with the addition of an aluminium block on top of the absorber in order to add weight. The second attempt was successful and this can be seen in Fig. 6 where the integrated noise power time series during the calibration process is shown. Liquid nitrogen was added twice, as the first time the liquid boiled and evaporated within minutes due to the addition of the aluminium block. The results in Fig. 6 indicate that an approximate black body target transitioning from 280K to 77K results in a differential integrated noise power level of 0.45dB.

### B. Calibration

The tip-curve calibration described in Section II was performed during a clear sky day. The receiver antenna elevation angle was tipped at elevation angles of 15°, 20°, 30°, 45°, 60° and 90° and the channel noise was recorded. The

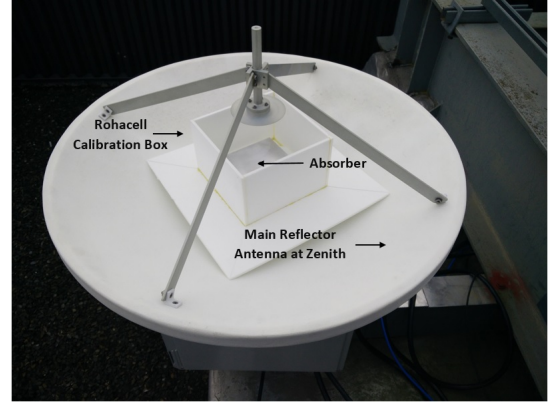


Fig. 5. Reflector at zenith to accommodate the calibration box during the hot-cold experiment.

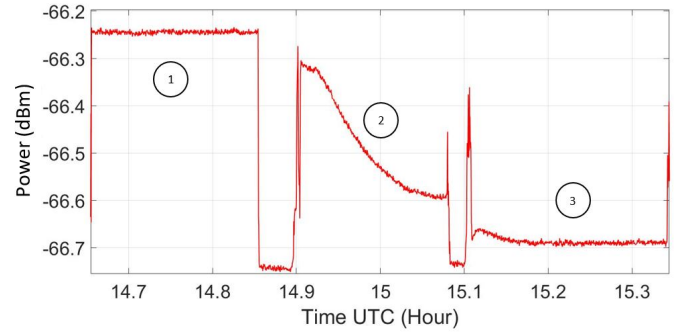


Fig. 6. Noise power recorded during the hot-cold target experiment. Different phases are marked as: (1) Noise Power with Absorber at Air Temperature. (2) Noise Power with Absorber immersed in Liquid Nitrogen (Failed attempt). (3) Noise Power with Absorber immersed in Liquid Nitrogen (Successful attempt).

ambient temperature target measurements recorded in the hot-cold experiment were used in the first approximation of the coefficients and attenuation. Fig. 8 shows the results of the initial and corrected calibration coefficients.  $\Delta T$  was adjusted to satisfy the assumption of no attenuation at an airmass of zero, the corrected coefficients are obtained as:

$$a = 8.98 * 10^{12} K J^{-1} Hz^{-1}, b = -1841 K$$

Furthermore, receiver gain variations due to temperature can impact the accuracy of the radiometer and can be compensated by performing a temperature calibration. To combat gain variation, in the work of [28] and [29], a reference noise source was added between the antenna and low noise block via a coupler. Due to hardware limitations, such calibration has not been performed. Nonetheless, to mitigate such errors, as mentioned in Section III the receiver is thermally insulated. Furthermore, the temperature variation of the low noise amplifier was observed to be stable with  $\pm 2^\circ C$  variations from the mean temperature.

The corrected coefficients obtained from the tip-curve calibration have been used to obtain the brightness temperature in the rest of the work.

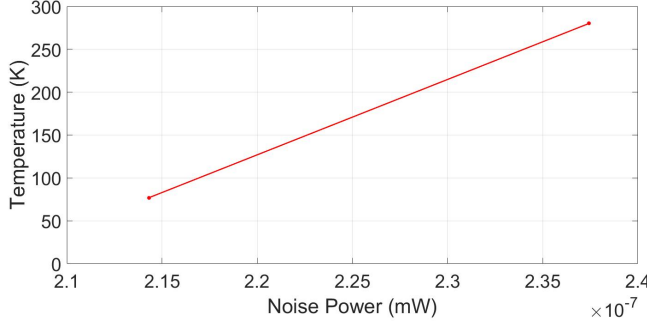
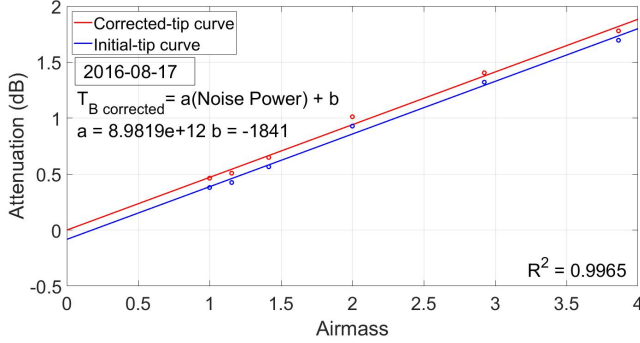


Fig. 7. Hot-Cold experiment plot.


 Fig. 8. Tip curve calibration, corrected-tip in red line, initial-tip in blue line. Coefficient of determination ( $R^2$ ) is 0.9964.

#### IV. EXPERIMENTAL RESULTS

The calibration performed in Section III is next applied to the measurement of atmospheric fading at Q-band. In order to enable concurrent measurements from the beacon and radiometric signals, the receiver was pointed to the geostationary ALPHASAT satellite.

In order to confirm the efficacy of the digital filtering in suppressing beacon signal power from being injected into the integrated noise, an experiment is conducted. In particular, measurements of the integrated noise are recorded when pointing the receiver at two different positions; while pointing to the ALPHASAT satellite and while pointing at a small angle away from the satellite during clear sky. The results are plotted in Fig. 9, where the measurements taken at the two positions of the receiver are marked. As shown, there is a small level of additional noise (approximately 0.03dB) recorded when the receiver points to the beacon. This increase is attributed to leakage of the beacon signal into the integrated noise.

In order to remove this contribution, the additional noise source was modelled as a temperature increase [13] as the sun's radiation in [30] was modelled as a contribution to brightness temperature. Furthermore, considering that the temperature and power are linearly related:

$$P_N(\alpha) = P_N(\text{away}) + P_N(\text{sat})e^{-(Att/4.343)} \quad (15)$$

where,  $P_N(\alpha)$  is the noise power measurement when tracking the satellite beacon,  $P_N(\text{away})$  is the noise power measurement when pointing away from the satellite beacon,  $P_N(\text{sat})$  is the satellite signal noise contribution in the absence

TABLE I  
RADIOMETER ATMOSPHERIC ABSORPTION IN CLEAR SKY THROUGHOUT CLEAR SKY SEGMENT

Hour	Temp ( $^{\circ}\text{C}$ )	$T_B$ (K)	$T_m$ (K)	$A_{ab}$ (dB)	Beacon (dBm)
1	10.90	72.49	267.42	1.33	-64.65
2	10.00	71.99	266.69	1.32	-64.61
3	9.42	71.76	266.22	1.32	-64.60
4	8.74	72.14	265.67	1.33	-64.61
5	8.55	68.11	265.51	1.24	-64.57
6	8.62	66.86	265.57	1.22	-64.52
7	9.75	62.89	266.50	1.12	-64.46
8	10.91	59.69	267.43	1.05	-64.38

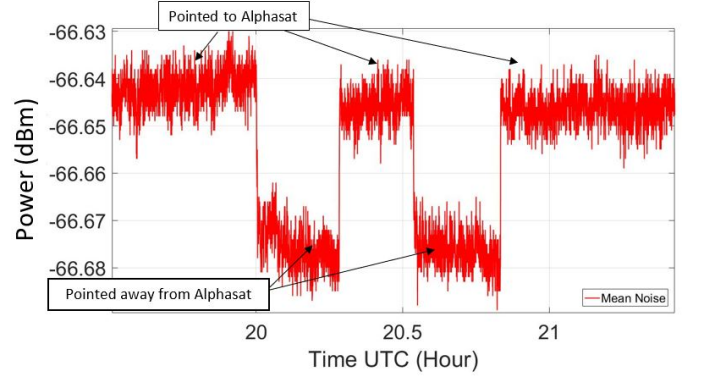


Fig. 9. Noise power recorded when pointed towards and away from ALPHASAT Aldo Paraboni beacon.

of atmospheric attenuation and  $Att$  is the path attenuation. The path attenuation,  $Att$ , during the measurement period was obtained from the ECMWF data. Once this bias is obtained,  $P_N(\text{sat})$ , is subtracted from the noise power measurements.

With noise bias from the beacon removed, estimations of the brightness temperature of the sky can be calculated from the integrated noise with the described method in the previous section. To check for any gain variations caused from temperature fluctuations, the estimated brightness temperature and received beacon power were plotted during a clear sky segment, Fig. 10. Mild temperature fluctuations during clear sky are expected to have minor changes on the estimated brightness temperature and received beacon power assuming no gain variations. The estimated brightness temperatures,  $T_B$ , mean brightness temperature,  $T_m$ , atmospheric absorption,  $A_{ab}$  and beacon signal at the different hours are reported in Table I. Reported temperatures in Fig. 10 (orange) and Table I from the nearby weather station were used to calculate  $T_m$  using (3) and  $A_{ab}$  was calculated using (2). A fluctuation of  $T_B$  and therefore  $A_{ab}$  can be seen in the period between hour 4 to 8 in which the temperature increased  $2.4^{\circ}\text{C}$ . The difference in atmospheric absorption between hour 1 and 5 is 0.09dB which can also be seen at the beacon signal as well. This indicates that the radiometer is operating as expected with no noticeable gain variations since integrated noise power and brightness temperature decreases with an increasing received beacon power.

The noise derived attenuation shown in Fig. 11 (in red) for a period of 26 days in July 2016 is obtained by the method described previously. For comparison, Fig. 11 (in

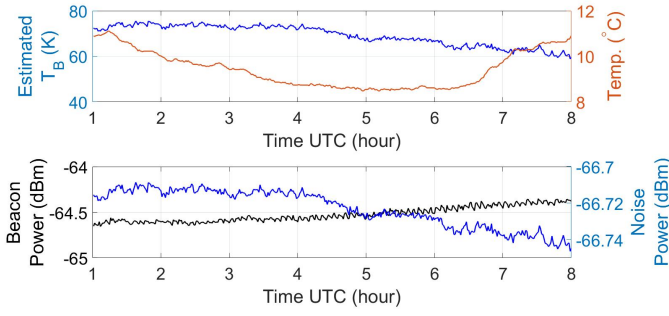


Fig. 10. Estimated brightness temperature (blue) with temperature (orange) superimposed (top) and received beacon power (black) with integrated noise power (blue) superimposed (bottom) during a clear sky segment.

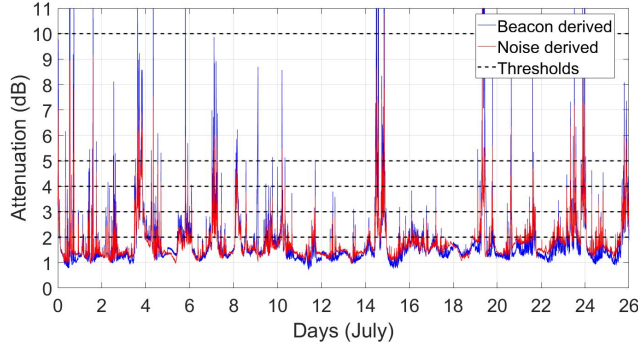


Fig. 11. Atmospheric attenuation for July 2016. In blue the atmospheric attenuation derived through power measurements, in red derived through the noise power measurements and in black different thresholds.

blue) also plots the beacon derived fading over the same period, which for the rest of the work is assumed to be the reference fading value. It should be noted that the beacon and noise power measurements were averaged at 1-minute intervals prior to the derived attenuations. The basis of averaging the measurements was to improve the resolution of the radiometer and the accuracy of the beacon power measurements. Also, the mean radiating temperature,  $T_m$ , was calculated as previously described. Consequently, the error of the radiometric measurement is next quantified by the difference from the beacon derived measurement. A first visual inspection indicates an overall good agreement between the two curves at least for low fading values. In order to further quantify this observation, Fig. 12 plots the difference (in dB) between the two sets of measurements. The mean value of this curve is 0.18dB with a standard deviation of 0.13dB. Moreover, the Cumulative Distribution Function (CDF) curve of this data shows values less than 1dB, 97.0 % of the time. This indicates that the probability to obtain a radiometric measurement of the fading with error less than 1dB is 0.97. Similar analysis indicates that the radiometric measurement can offer better accuracy than 0.5dB with a probability of 0.937.

Since it is theoretically expected that the radiometric measurements would be more accurate for the low fading regime, the accuracy of the radiometric estimations is next quantified when measurements over a certain fading threshold are excluded. In particular, thresholds of fading at 10dB, 5dB, 4dB,

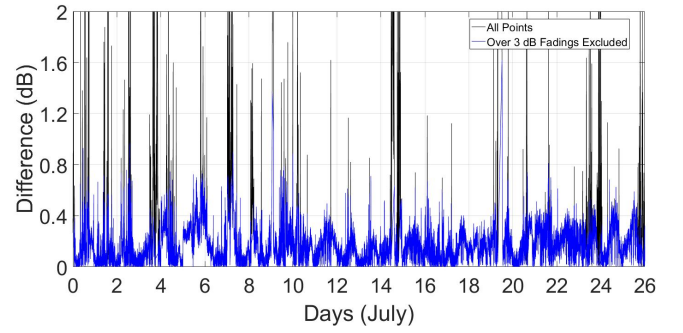


Fig. 12. Beacon derived and noise derived attenuation difference with all points included (black) and with over 3dB fadings excluded (blue).

3dB and 2dB are considered. The difference (in dB) between the two sets of measurements for a threshold are obtained and any values exceeding the set threshold, are not considered. Fig. 12 (in blue) illustrates the difference (in dB) between the radiometric and beacon derived attenuations for a threshold value of 3dB.

It is noted that in practice, the low fading regime is of particular interest for the operation of satellite feeder links in the Q/V-band and beyond. This is due to the limited offerings in power amplification technologies at these frequencies [31], which are likely to limit the fade margins of the gateway links compared to lower frequency bands. Indeed in place of large fade margins, Q/V-band feeder links increasingly rely on site diversity [32] as means to combat fading.

Consequently, a similar analysis is performed for the different threshold values. The CDF for the difference between the two sets of measurements when different threshold values are considered are plotted in Fig. 13. Table II provides some summary results indicating the probability as a percentage for the radiometric measurement to deliver valid result within error margins of 0.5dB and 1dB for different fading estimation ranges. As shown, when targeting to obtain fading readings up to 5dB, the probability of the error to be below 1dB is 0.991. This probability reduces to 0.976 when fading up to 10dB is targeted. It is noted that although in this study the beacon derived fading is considered as reference, in practice there is some uncertainty also with this approach. This is indicated by e.g. the standard deviation of the beacon derived fading over typical bright sky conditions being of the order of 0.25dB. Moreover, calibration of any gain variation from the receiver would yield improved results and strengthen the accuracy for the use of a standalone radiometer for monitoring fading.

## V. DISCUSSION AND CONCLUSION

A Q-band SDR based terminal installed at Heriot-Watt University receiving the Aldo Paraboni beacon was used to evaluate the potential of utilizing digital noise power integration as an estimate for passive radiometry. Radiometric measurements of the atmosphere were derived from the calibrated integrated noise and the results were compared against concurrent beacon derived measurements. The beacon and radiometric derived measurements indicated a good agreement particularly in the low fading regime. Indicatively, and assuming the beacon



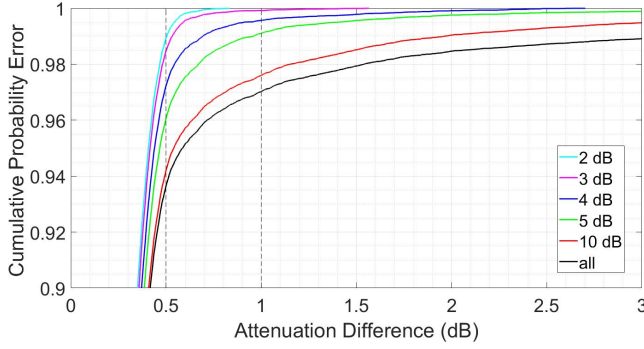


Fig. 13. CDF of the attenuation difference between the beacon derived and noise derived attenuation for July 2016 including all points (black) and the shown thresholds.

TABLE II  
THE CDF OF  $\leq 0.5\text{dB}$  AND  $\leq 1\text{dB}$  FOR ALL POINTS AND DIFFERENT FADING THRESHOLDS

Fading Thresholds	CDF	
	$\leq 0.5\text{dB}$	$\leq 1\text{dB}$
None	93.7%	97.0%
$\leq 10\text{dB}$	94.2%	97.6%
$\leq 5\text{dB}$	96.1%	99.1%
$\leq 4\text{dB}$	97.2%	99.6%
$\leq 3\text{dB}$	98.5%	99.9%
$\leq 2\text{dB}$	99.0%	100.0%

derived measurement as reference, the error probability of the SDR radiometer delivering error  $>1\text{dB}$  for fadings up to  $5\text{dB}$  is of the order of 1%. This range and resolution can be suitable for the design of future Q/V-band gateways and beyond, which due to technology constraints, are likely to operate at lower fade margins compared to existing systems up to Ka-band. However in principle the proposed setup and calibration methodology can achieve improved resolution on the basis of hardware with superior performance and in this case be exploited during the operation of a high frequency link. According to prevailing standards [33], propagation impairment mitigation techniques (PIMT) such as adaptive coding and modulation will be deployed with a very fine granularity for the fading. Indicatively we note that the entire spectrum of available coding rates for 16APSK modulation extends over signal to noise ratio of about  $6\text{dB}$  [33]. Consequently these findings reflect to potentially significant gains in the application of PIMT during deployment, while dispensing the need for additional costly instrumentation (e.g. dedicated radiometer) by virtue of enabling a built-in radiometric observations within the existing receiver hardware.

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